

- I imagine a 220 kg satellite in orbit 640 km above the Earth.

- What are initial speed & period?

$$mv^2/r = GMm/r^2$$

$$\Rightarrow v^2 = GM/r$$

$$\Rightarrow v = \sqrt{GM/r}$$

$$= \sqrt{6.7 \times 10^{-11} \cdot 6 \times 10^{24} / (6378000 + 640000)}$$

$$\sim \sqrt{40 \times 10^{13} / 7 \times 10^6}$$

$$\sim \sqrt{6 \times 10^7}$$

$$\sim 8000 \text{ m/s}$$

$$= \boxed{8 \text{ km/s}}$$

$$T = 2\pi r/v = \frac{2\pi \cdot 7 \times 10^6}{8000}$$

$$\sim \frac{40 \times 10^6}{8 \times 10^3} = 5000 \text{ s}$$

$$\equiv \boxed{90 \text{ min}}$$

- If satellite loses mechanical energy of $1.4 \times 10^5 \text{ J}$ per orbit, but maintains a circular orbit, what happens after 1500 orbits?

In circular orbit $K = \frac{GMm}{2r}$
 $= -U/2$

$$K_{\text{init}} = \frac{1}{2} \cdot 220 \cdot 8000^2$$
$$= 110 \cdot 64 \times 10^6$$
$$\approx 7 \times 10^8 \text{ J}$$

$$\Delta K = 1.4 \times 10^5 \cdot 1500$$
$$= 2100 \times 10^5$$
$$= 2.1 \times 10^8$$

$$K_{\text{final}} = K_{\text{init}} + \Delta K$$
$$= 7.2 \times 10^8$$

$$\frac{r_{\text{final}}}{r_{\text{init}}} = \frac{K_{\text{init}}}{K_{\text{final}}} \approx 0.97$$

altitude $\approx 420 \text{ km}$

V increases slightly
and T decreases slightly

- What is magnitude of retarding force?

$$W = \Delta KE \\ = F \Delta s$$

$$\Delta s = 2\pi r \\ \approx 40 \times 10^6 \text{ m}$$

$$\Delta KE = 1.4 \times 10^5$$

$$\Rightarrow F = \Delta KE / \Delta s \\ = \frac{1.4 \times 10^5}{4 \times 10^7}$$

$$= \boxed{3.5 \times 10^{-3} \text{ N}}$$

- Very small drag force will deorbit a satellite

$$1500 \text{ orbits @ } 90 \text{ min/orbit} \\ \Rightarrow 135000 \text{ min} \\ = 2250 \text{ hr} \\ < 100 \text{ days}$$

- In one-dimension:

$$\Delta U_{ab} = - \int_a^b F(x) dx$$

With inverse operation

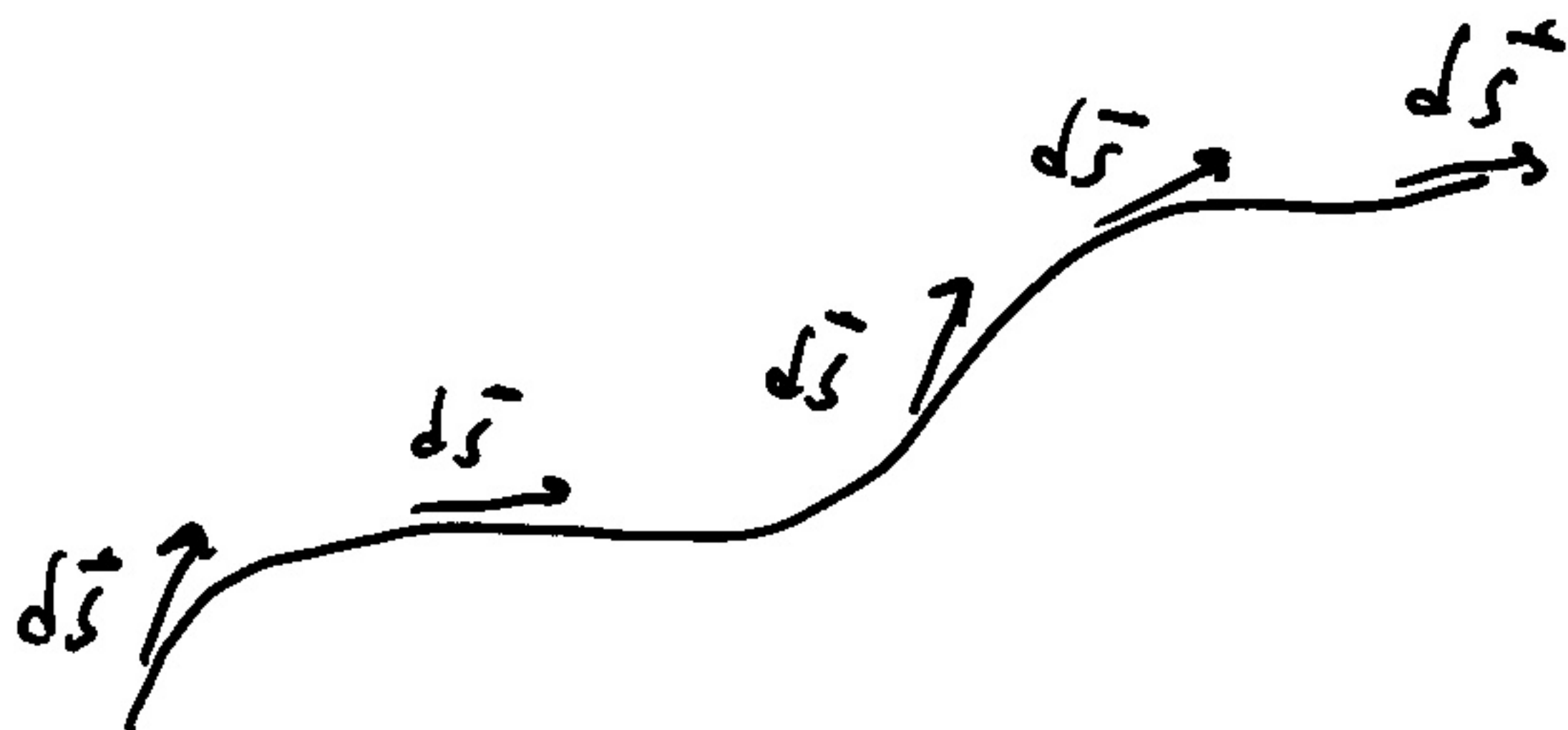
$$F(x) = - \frac{dU}{dx}$$

- In three-dimensions:

We have seen that

$$\Delta U = - \int_a^b \vec{F}(\vec{r}) \cdot d\vec{s}$$

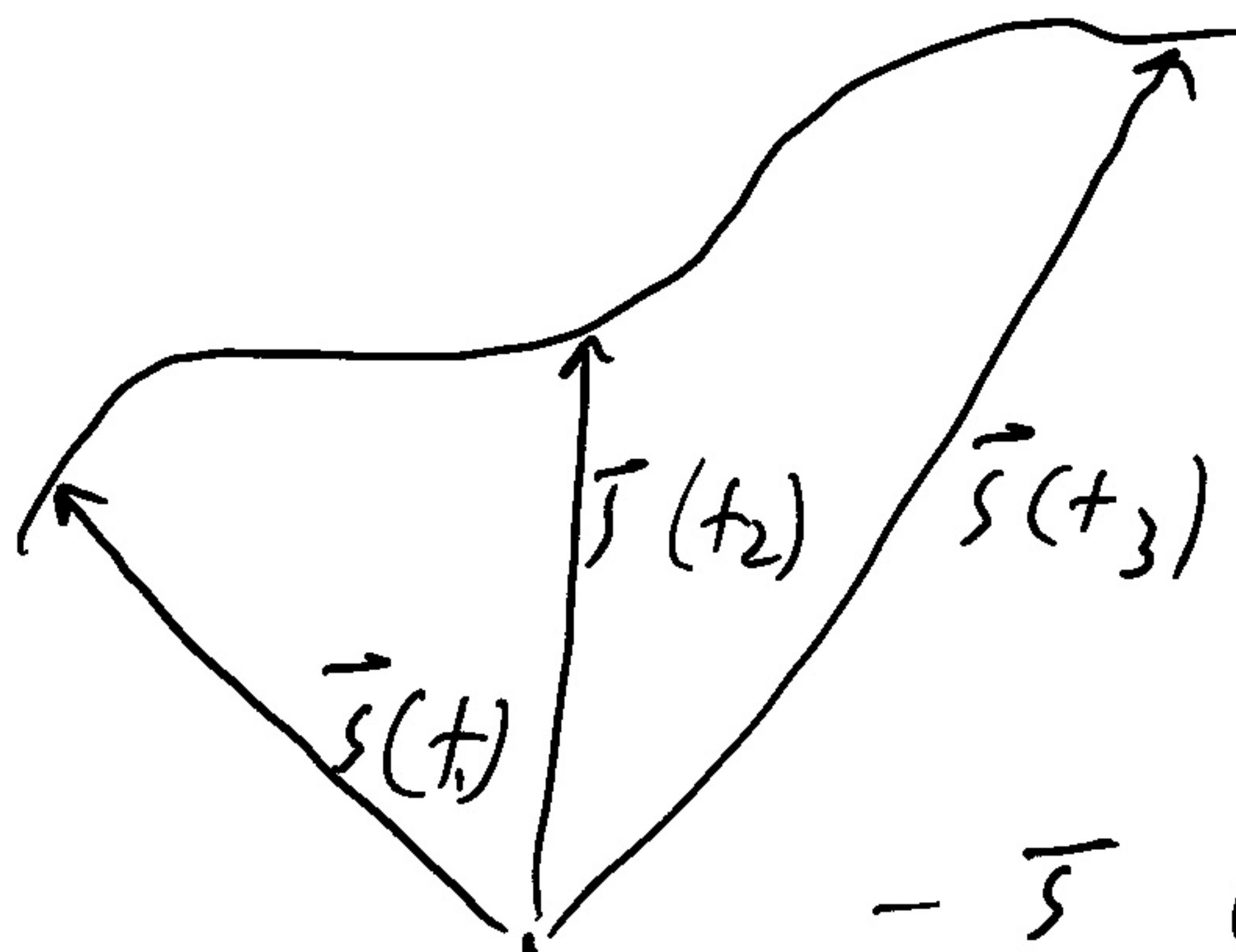
where $d\vec{s}$ is tangent to a curve



What is the inverse?

$$\vec{F}(\vec{r}) = - \nabla U = - \left[\frac{\partial U}{\partial x}, \frac{\partial U}{\partial y}, \frac{\partial U}{\partial z} \right]$$

First let's parameterize our curve as a function of a dummy variable t



- \vec{s} is a position vector

$\vec{s}'(t)$ is a velocity vector tangent to the curve

$$\vec{s}'(t) = \left[\frac{dx}{dt}, \frac{dy}{dt}, \frac{dz}{dt} \right]$$

where $\vec{s}(t) = [x(t), y(t), z(t)]$

$$\text{so } - \int \vec{F}(\vec{r}) \cdot d\vec{s}$$

$$= - \int \vec{F}(\vec{s}(t)) \cdot \vec{s}'(t) dt$$

$$-\int \vec{F}(\vec{s}(t)) \cdot \vec{s}'(t) dt$$

$$= -\int [F_x \frac{dx}{dt} + F_y \frac{dy}{dt} + F_z \frac{dz}{dt}] dt$$

Need this to be
a perfect differential

Use chain rule on $U(\vec{r})$

$$dU(\vec{r})/dt = dU(x(t), y(t), z(t))/dt$$

$$= \frac{\partial U}{\partial x} \frac{dx}{dt} + \frac{\partial U}{\partial y} \frac{dy}{dt} + \frac{\partial U}{\partial z} \frac{dz}{dt}$$

$$= \left[\frac{\partial U}{\partial x}, \frac{\partial U}{\partial y}, \frac{\partial U}{\partial z} \right] \cdot \vec{s}'(t)$$

$$= \nabla U \cdot \vec{s}'(t)$$

$$\int_a^b \frac{dU}{dt} dt = U(b) - U(a)$$

$$= \Delta U_{ab}$$

$$= \int_a^b \nabla U \cdot \vec{s}'(t) dt$$

$$= \int_a^b \nabla U \cdot d\vec{s} = - \int_a^b \vec{F} \cdot d\vec{s}$$

$$\Rightarrow \vec{F} = -\nabla U$$

$$\nabla U = \text{grad } U$$

= gradient of U

- What is physical meaning?
- Just like dy/dx in 1-d tells you the slope in 1-d, ∇U tells you the slope in 3-d.

$\frac{\partial U}{\partial x}$ is slope in x -direction
 $\frac{\partial U}{\partial y}$ " " y -direction
 $\frac{\partial U}{\partial z}$ " " z -direction

~ Where does ∇U point?

It points uphill!

so $-\nabla U = \vec{F}$ points downhill